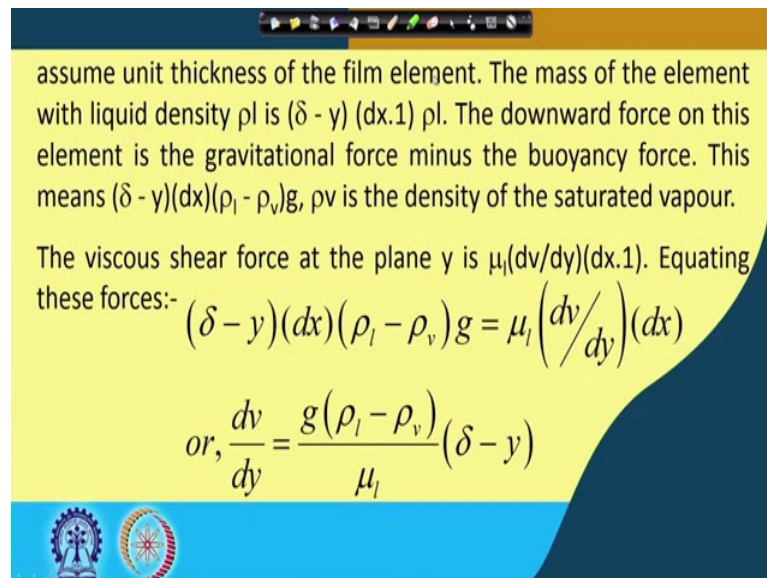


**Thermal Operations in Food Process Engineering: Theory and Applications**  
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**Lecture - 42**  
**Condensation (Contd.)**

Good morning. So, in that Condensation class, the last class we had come to the point that  $dv/dy$  was  $(\rho_l - \rho_v) / (\delta - y)$  and that expression, 'right'. And we will start from there ok. So, this is our condensation continued class and lecture number 42, this is sorry lecture number 42, 'right', which we can underline like this, ok.

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assume unit thickness of the film element. The mass of the element with liquid density  $\rho_l$  is  $(\delta - y)(dx) \rho_l$ . The downward force on this element is the gravitational force minus the buoyancy force. This means  $(\delta - y)(dx)(\rho_l - \rho_v)g$ ,  $\rho_v$  is the density of the saturated vapour.

The viscous shear force at the plane  $y$  is  $\mu_l(dv/dy)(dx)$ . Equating these forces:-

$$(\delta - y)(dx)(\rho_l - \rho_v)g = \mu_l \left( \frac{dv}{dy} \right) (dx)$$
$$\text{or, } \frac{dv}{dy} = \frac{g(\rho_l - \rho_v)}{\mu_l} (\delta - y)$$

So, in lecture number 42, we will start with that the expression which we had given earlier as this, 'right' which we had given earlier as this that  $dv/dy$  is  $g \times (\rho_l - \rho_v) / \mu_l \times (\delta - y)$ .

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where,  $\delta$  is the thickness of the condensate layer at the position  $x$ ,  $\mu$  is the viscosity, and subscripts  $l$  and  $v$  refer to the liquid and vapour phase respectively. We are assuming in this case zero shear stress at the liquid vapour interface because vapour is assumed to be stationary. At the wall surface liquid velocity is zero.  $\therefore v = 0$  at  $y = 0$ . Integrating and using this boundary condition, the velocity distribution in the condensate layer is

$$v(y) = \frac{g(\rho_l - \rho_v)}{\mu_l} \left( \delta y - \frac{y^2}{2} \right)$$

The mass flow rate of condensate  $m(x)$  through any axial position  $x$  per unit depth of the plate is

Start from there of course, the individual terms we know new term came up with that is the  $\delta$  is the thickness of the condensate layer at the position  $x$  at any position if you remember we said that this was wall and this was the flow of the condensate, 'right' at any position  $x$ , 'right', this is the  $x$  and  $y$  direction, this is  $y$ , this is  $x$ . So, at any position  $x$ , 'right', the thickness is  $\delta$  that is what we are saying the thickness is  $\delta$ . And  $\mu$  is the viscosity of the liquid and subscripts  $l$  and  $v$  refers to liquid and vapor phase respectively, 'right'.

So, and we are assuming in the case of zero shear stress, 'right' that is  $\tau = 0$ , in the case of zero shear stress at the liquid vapor interface, what is the liquid vapor interface? Liquid vapor interface is this, 'right', this is that because this side it is vapor, this is liquid which is coming in contact with this cold surface, 'right'. So, this is vapor.

So, at this liquid vapor interface say at  $x$ , so we can take it here, 'right'. So, at the liquid vapor interface like if we take that shear stress is zero, 'right' and this is assumed to be a stationary. So, we are assuming in this case that zero shear stress at the liquid vapor interface because vapor is assumed to be stationary at that point or as such. At the wall surface liquid velocity is zero, so,  $v$  is 0 again this is the wall, 'right' this was our fluid or the condensate.

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where,  $\delta$  is the thickness of the condensate layer at the position  $x$ ,  $\mu$  is the viscosity, and subscripts  $l$  and  $v$  refer to the liquid and vapour phase respectively. We are assuming in this case zero shear stress at the liquid vapour interface because vapour is assumed to be stationary. At the wall surface liquid velocity is zero.  $\therefore v = 0$  at  $y = 0$ . Integrating and using this boundary condition, the velocity distribution in the condensate layer is

$$v(y) = \frac{g(\rho_l - \rho_v)}{\mu_l} \left( \delta y - \frac{y^2}{2} \right)$$

The mass flow rate of condensate  $m(x)$  through any axial position  $x$  per unit depth of the plate is

So, the liquid which is at the wall so we normally say that it is clinging to the surface clinging it is clinging to the surface, 'right' attached fully, 'right'. It is now allowing to move, so that is what we call clinging to the surface, 'right'. So, if that be true, then the velocity  $v$  is 0 at  $y$  is 0, which one is  $y$ ?

Again the wall so this was the fluid or condensate and we said this was  $x$ -direction this was  $y$ -direction. So,  $y$  is 0 is this, 'right'. So,  $v$  is 0 at  $y$  is 0, then that equation which just now we had shown that equation which just now we had shown, 'right' like this one that  $dv/dy = g(\rho_l - \rho_v) / \mu_l \times (\delta - y)$ .

So, if we look at this equation, if we integrate that equation/ 'right' with the boundary condition that boundary condition is what  $v = 0$  at  $y = 0$ , then the velocity distribution in the condensate that can be written as  $v(y) = g(\rho_l - \rho_v) / \mu_l \times (\delta y - (y^2 / 2))$ , 'right'.

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where,  $\delta$  is the thickness of the condensate layer at the position  $x$ ,  $\mu$  is the viscosity, and subscripts l and v refer to the liquid and vapour phase respectively. We are assuming in this case zero shear stress at the liquid vapour interface because vapour is assumed to be stationary. At the wall surface liquid velocity is zero.  $\therefore v = 0$  at  $y = 0$ . Integrating and using this boundary condition, the velocity distribution in the condensate layer is

$$v(y) = \frac{g(\rho_l - \rho_v)}{\mu_l} \left( \delta y - \frac{y^2}{2} \right)$$

The mass flow rate of condensate  $m(x)$  through any axial position  $x$  per unit depth of the plate is

From there we can say that the equation which we had, it was  $dv/dy$  and it was  $(\rho_l - \rho_v)$ , 'right' over  $\mu_l$ , 'right' into  $\delta \times g$ , 'right' times  $(\delta - y)$ , 'right',  $(\delta - y)$ , this was our expression, 'right'. So, when we integrated this, 'right', when we integrated this we have integration constant  $c$ , 'right' and the value of the integration constant and subsequently using that we get the velocity distribution  $v(y)$  is  $g(\rho_l - \rho_v) / \mu_l \times (\delta y - (y^2 / 2))$ , 'right', because it was  $dv/dy$ .

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where,  $\delta$  is the thickness of the condensate layer at the position  $x$ ,  $\mu$  is the viscosity, and subscripts l and v refer to the liquid and vapour phase respectively. We are assuming in this case zero shear stress at the liquid vapour interface because vapour is assumed to be stationary. At the wall surface liquid velocity is zero.  $\therefore v = 0$  at  $y = 0$ . Integrating and using this boundary condition, the velocity distribution in the condensate layer is

$$v(y) = \frac{g(\rho_l - \rho_v)}{\mu_l} \left( \delta y - \frac{y^2}{2} \right)$$

The mass flow rate of condensate  $m(x)$  through any axial position  $x$  per unit depth of the plate is

So,  $dy$  was integrated. So, this side it would have been  $y^2 / 2$  that is what is coming, 'right'. And this is nothing, but; this is nothing but expression for a parabolic flow of the fluid. So,  $v(y)$  is a parabolic in nature and that is how it is coming like this, 'right'. So; the mass flow rate of the condensate  $m(x)$  through any axial position  $x$  per unit depth of the plate that can be written as.

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$$m(x) = \int_0^{\delta} \rho_l v dy$$

$$= \int_0^{\delta} \rho_l \frac{g(\rho_l - \rho_v)}{\mu_l} \left( \delta y - \frac{y^2}{2} \right) dy$$

$$\therefore m(x) = \frac{g \rho_l (\rho_l - \rho_v) \delta^3}{3 \mu_l} \dots (M)$$

This can be written as  $m(x) = \int_0^{\delta} \rho_l v dy$  'right'  $\int_0^{\delta} \rho_l v dy$ , 'right', which we can rewrite is equal to  $\int_0^{\delta} \rho_l v$  already got in this form  $g \times (\rho_l - \rho_v) / \mu_l$  into  $(\delta y - (y^2/2)) \times dy$ , 'right'. So,  $dy$  was there,  $v$  we substituted,  $\rho_l$  we take took and this is between 0 to 1 integration, 'right'.

So, it is a definite integral and we get  $m(x) = g \rho_l \times (\rho_l - \rho_v) \times \delta^3 / 3 \mu_l$  which we turn to with the equation  $m$ , 'right'. So, we got the expression  $m(x)$  for in terms of  $\delta$  and rho's in terms of del rho's and  $g$  we got the expression for  $m(x)$ .

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At the wall for area  $(dx \cdot 1) \text{ m}^2$ , the linear temperature distribution is assumed and the heat transfer becomes

$$q_x = -k_l(dx \cdot 1) \frac{dT}{dy} \Big|_{y=0} = k_l dx \frac{T_{sat} - T_w}{\delta}$$

In a  $dx$  distance, the rate of heat transfer is  $q_x$ . Also, in this  $dx$  distance, the increase in mass from condensation is  $dm$ . From equation (M) we can write

$$dm = d \left[ \frac{g \rho_l (\rho_l - \rho_v) \delta^3}{3 \mu_l} \right] = \frac{g \rho_l (\rho_l - \rho_v) \delta^2 d\delta}{\mu_l}$$

At the wall for area  $dx$  times  $1$  at the wall for the area  $dx$  times  $1 \text{ m}^2$ , the linear temperature distribution is assumed and the heat transfer becomes, because we have said that  $dx$  times  $1$  that is the wall area, 'right'. So, this was wall and this is  $x$  and the other side this side is taken unity, 'right'. So, that means we can say that this unit is like that.

And here we say that in this area, the linear temperature distribution is assumed and the heat transfer then we can write  $q_x = -k dx \times 1$  that is the area  $dT dy$  at the position  $y = 0$ , 'right'. So, this we can rewrite that this is  $k_l k_l dx \times (T_{saturation} - T_w) / \delta$  this  $dT dy$  is  $T_{saturation}$  at the point  $y = 0$ . I repeat where the point  $y = 0$  is this point, 'right'.

So, there we are saying that this is  $T_{saturation} dT$  means  $(T_{saturation} - T_w)$  since  $y$  is  $0$ . So, we can write here it is equal to  $\delta$ , 'right'. And since it would have been  $T_w - T_{saturation}$ , but with the negative, so  $(T_{saturation} - T_w)$ , 'right', so it is  $k_l dx (T_{saturation} - T_w) / \delta$ .

In a  $dx$  distance that is vertical this was our  $x$  direction so in a  $dx$  distance the rate of heat transferred is  $q_x$ . Also, in this  $dx$  distance the increase in mass from the condensation is  $dm$ , 'right', because we saw that this curve was like that. So, whatever mass was here mass is more at this place and mass is more at this place, so it is gradually changing, 'right'. So, we are saying in a  $dx$  distance, the rate of heat transfer  $q_x$ . Also, in this  $dx$  distance, the increase in mass from the condensation is  $dm$ , 'right'.

So, earlier equation which we wrote as  $m$  we can rewrite that  $dm$  is equal to  $d$  of  $m$ ;  $m$  is  $g \times \rho_l \times (\rho_l - \rho_v) \times \delta^3 / 3 \mu_l$ , 'right'. So, this we can write that  $g (\rho_l - \rho_v)$  into since it is  $d$  of that  $dm$  of this, then this becomes  $\delta^2 d \delta / \mu_l$ , 'right', differentiating  $d$  with  $\delta$  is a  $\delta$  if we differentiate it becomes  $\delta^2 d \delta$ , on integration this gives  $\delta^3$  'right'. So, this is  $\delta^2 d \delta$  is  $dm$ , 'right'. So, again we got equation which we need to integrate, 'right'.

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Where,  $dm$  represents the rate of condensation over the distance  $dx$  per unit depth of the plate, since the condensate thickness increases by  $d\delta$  over the differential length  $dx$ . The rate of heat released  $dQ$  associated with the rate of condensation  $dm$  is  $dQ = h_{fg} dm \dots (H)$ ; where  $h_{fg}$  is the latent heat of condensation. The amount of heat released  $dQ$  over the area  $dx \times 1$  must be transferred across the condensate layer of thickness  $\delta$  by conduction which we have assumed in assumption no 6.

$$\therefore dQ = k_l \frac{T_{sat} - T_w}{\delta} dx \cdot 1$$

Where,  $dm$  is the representation of the rate of condensation over the distance  $dx$  per unit depth of the plate, 'right', again unit depth is that third dimension which you are referring to. So, this is  $y$  and this is that third dimension is unit, this is  $x$ , 'right', so that we are referring to.

Since the condensate thickness since the condensate thickness not  $b$ , since the condensate thickness increases by  $d\delta$  over the differential length of  $dx$ , the rate of heat released  $dQ$  associated with the rate of condensation  $dm$  is  $dQ$  equal to  $h_f g$  into  $dm$ , 'right'. This we write with the equation  $H$ . Of course,  $h_f g$  has to be said where  $h_f g$  is the latent heat of condensation, 'right' this we have we also used in the previous class  $h_f g$  for latent heat of condensation or vaporization whatever we call.

The amount of heat released  $dQ$  over the area  $dx$  times 1 must be transferred across the condensate layer of  $\delta$  thickness by conduction which we have assumed, 'right' in the assumption number 6, 'right'. So, therefore, we can write that  $dQ$  is  $k_l (T_{saturation} - T_w) / \delta$  times the area that is  $dx \times 1$ , 'right'. So, we got also  $dQ$  in terms of  $dx$ , 'right'.

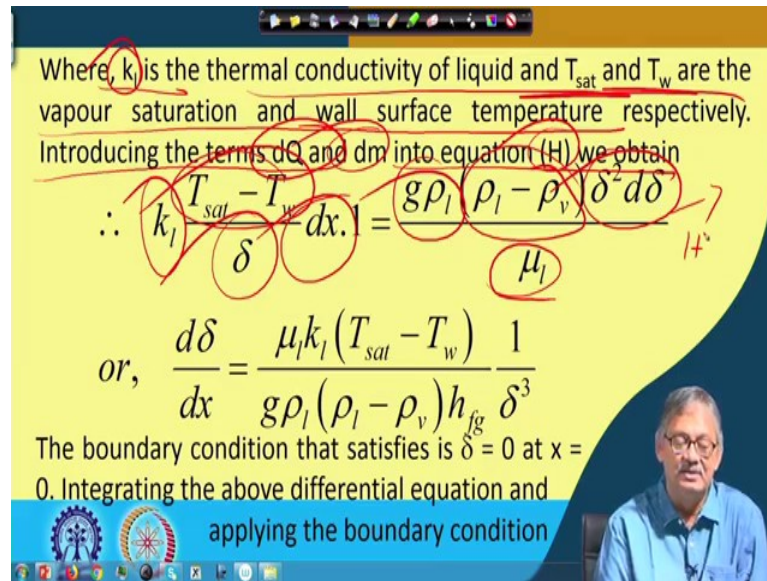
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Where,  $k_l$  is the thermal conductivity of liquid and  $T_{sat}$  and  $T_w$  are the vapour saturation and wall surface temperature respectively. Introducing the terms  $dQ$  and  $dm$  into equation (H) we obtain

$$\therefore k_l \frac{T_{sat} - T_w}{\delta} dx \cdot 1 = \frac{g \rho_l (\rho_l - \rho_v) \delta^2 d\delta}{4 \mu_l}$$

or, 
$$\frac{d\delta}{dx} = \frac{\mu_l k_l (T_{sat} - T_w)}{g \rho_l (\rho_l - \rho_v) h_{fg} \delta^3}$$

The boundary condition that satisfies is  $\delta = 0$  at  $x = 0$ . Integrating the above differential equation and applying the boundary condition



So, we can say of course, the new term which has come up in this is the  $k$  that is the  $k_l$  is the thermal conductivity of the liquid,  $T_{sat}$  and  $T_w$  we have said earlier also are the vapor saturation and walls surface temperature respectively, 'right'. So, introducing the term  $dQ$  and  $dm$  into the equation earlier we have defined as  $H$  we can write that  $k_l (T_{saturation} - T_w) / \delta$  times the area  $dx \times 1$  that is equal to  $g \rho_l \times (\rho_l - \rho_v) \times \delta^2 \delta / 4 \mu_l$ , this equation we wrote as equation H, 'right'.

So, we can rewrite this equation as  $d \delta / dx$  because we have here  $\delta$  and we have  $dx$ . So, we can rewrite that  $d \delta / dx$  is equal to  $\mu_l k_l \times (T_{sat} - T_w) / g \times \rho_l \times (\rho_l - \rho_v) \times h_{fg}$ , 'right' times  $1/\delta^3$ , 'right'. And the boundary form because this is if we want to integrate we need a boundary.

So, that boundary condition that satisfies is  $\delta = 0$  at  $x = 0$ , 'right',  $\delta = 0$  at  $x = 0$  which is what this was our wall and this was our that fluid which is condensing we said our  $y$  is in this direction, our  $x$  is in this direction.



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Where,  $k_l$  is the thermal conductivity of liquid and  $T_{sat}$  and  $T_w$  are the vapour saturation and wall surface temperature respectively. Introducing the terms  $dQ$  and  $dm$  into equation (H) we obtain

$$\therefore k_l \frac{T_{sat} - T_w}{\delta} dx \cdot 1 = \frac{g \rho_l (\rho_l - \rho_v) \delta^2 d\delta}{\mu_l}$$

or,  $\frac{d\delta}{dx} = \frac{\mu_l k_l (T_{sat} - T_w)}{g \rho_l (\rho_l - \rho_v) h_{fg} \delta^3}$

The boundary condition that satisfies is  $\delta = 0$  at  $x = 0$ . Integrating the above differential equation and applying the boundary condition

So, and since  $y$  is in that direction, we are taken that element, 'right' of thickness  $1$  and that was  $y \delta - y$ , 'right', this our thing. However, here since it is in the  $y$  direction, so at  $\delta = 0$  is  $x = 0$ , here it is  $\delta$  is  $0$  at  $x = 0$  that is the origin, 'right'. So, if we take that, then we can say the boundary condition that satisfies is this  $\delta$  is  $0$  at  $x = 0$  and integrating this equation and applying the boundary condition, we can write.

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we get the thickness of the condensate layer as a function of the position  $x$  along the plate as

$$\delta(x) = \left[ \frac{4 \mu_l k_l (T_{sat} - T_w) x}{g (\rho_l - \rho_v) \rho_l h_{fg}} \right]^{1/4}$$

Since we have established a relation for the thickness of the condensate layer, the local heat transfer coefficient  $h_x$  for condensation is determined from the heat balance as

$$h_x (dx \cdot 1) (T_{sat} - T_w) = k_l (dx \cdot 1) \frac{T_{sat} - T_w}{\delta(x)}$$

So, if we apply boundary and integrate, we can write that the thickness  $\delta$  that we can write in this form in this way that the thickness of the condensate layer is a function of

the position  $x$  along the plate that is also true. We have shown that this is the  $T_{\text{wall}} T_w$ , and this was the; this was the fluid which was condensing or liquidity is condensing, 'right'. So, if this is the  $x$  direction, so whatever thickness is here is not the same thickness here is not the same thickness here. So, this thickness is a function of  $x$ , 'right', so, that is what we are saying.

That  $\delta x = [4 \times \mu_l k_l \times (T_{\text{saturated}} - T_{\text{wall}}) \times x] / [g \times (\rho_l - \rho_v) \times \rho_l \times h_f g]^{1/4}$ , 'right', because it came that your  $\delta$  was cube and so when it make you it was integrated so it became 4 and that 4  $\delta$  4 has now become 1/4 in this power, 'right'. So, the thickness at any position  $x$  in that film is  $[4 \mu_l k_l (T_{\text{saturated}} - T_{\text{wall}}) \times x] / [g \times (\rho_l - \rho_v) \times \rho_l \times h_f g]^{1/4}$ .

Now, since we have established a relation for the thickness of the condensate layer, the local heat transfer coefficient  $h_x$  or condensation is determined from the heat balance as  $h_x \times dx \times 1 \times (T_{\text{saturated}} - T_{\text{wall}})$  is  $k_l \times dx \times 1 \times (T_{\text{saturated}} - T_{\text{wall}}) / \delta x$ , 'right'.

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or,  $h_x = \frac{k_l}{\delta(x)}$  Now, introducing the value of  $\delta(x)$  into this expression we get,

or,  $h_x = \frac{k_l}{\left[ \frac{4\mu_l k_l (T_{\text{sat}} - T_w) x}{g(\rho_l - \rho_v) \rho_l h_{fg}} \right]^{1/4}}$

$= \frac{[g\rho_l(\rho_l - \rho_v)h_{fg}k_l^3]^{1/4}}{4\mu_l(T_{\text{sat}} - T_w)x}$

So, if this be true, then by changing the value of  $\delta$  we can write that  $h_x = k_l \times a$  over  $\delta x$ , 'right'. So, this we can rewrite that now introducing the value of  $\delta x$  into this expression, we get  $h_x$  is  $k_l / [4/g \mu_l k_l (T_{\text{sat}} - T_{\text{wall}}) \times x] / [g \times (\rho_l - \rho_v) \times \rho_l \times h_f g]^{1/4}$  that is the  $h_x$  that is the heat transfer coefficient or at a localized heat transfer coefficient  $h_x$  at any position  $x$ . We had again reminded you that this was the direction of  $x$ .

So,  $h$  at position any position  $x$  is the localized heat transfer coefficient which can be rewritten  $k_1$  taking inside as  $\{[g \times \rho_l \times (\rho_l - \rho_v) \times h_{fg} k_l^3] / [4 \mu_l (T_{sat} - T_w) x]\}^{1/4}$ , 'right' this to the power 1/4 is the rewritten of this 'right'. This goes up, so  $4 \mu_l (T_{sat} - T_w)$ , 'right'  $4 \mu_l$  this was  $g \times (\rho_l - \rho_v) \times h_{fg} \times \rho_l h_{fg}$  of course  $\rho_l$  has been taken into that  $h_{fg}$  into this  $k_1$  that  $k_1$  has come down.

So, it is  $k_1$ , earlier we had 1  $k_1$  here to the power 1/4. So, when this  $k_1$  is going inside, so it has to be  $k_1^3/4 \mu_l (T_{sat} - T_w)$  and times  $k$  because  $k$  has been taken inside  $k_1$ , 'right'. So, this is the  $h_x$  the local heat transfer coefficient.

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Hence, the local heat transfer coefficient  $h_x$  varies with the distance  $x^{1/4}$ . The average heat transfer coefficient  $h_{av}$  over the length  $0 \leq x \leq L$  of the plate is

$$h_{av} = \frac{1}{L} \int_0^L h_x dx = \frac{4}{3} h_x |_{x=L}$$

$$= 0.943 \left[ \frac{g \rho_l (\rho_l - \rho_v) h_{fg} k_l^3}{4 \mu_l (T_{sat} - T_w) L} \right]^{1/4} \quad W/m^2 \cdot ^\circ C$$

For laminar flow, experimental data show that this result is 20% less than the experimental data.

Now, if we have the local heat transfer coefficient we can have an average heat transfer coefficient like this. The local heat transfer coefficient  $h_x$  varies with the distance  $x$  as  $x^{-1/4}$ . The average heat transfer coefficient  $h$  average over the length 0 to 1 is in the plate rather is  $h$  average is  $1/1 \int_0^1 h_x dx$  or it is  $4/3 h_x$  at any  $x = 1$  and that can be rewritten as 0.943 into  $[g \times \rho_l \times (\rho_l - \rho_v) h_{fg} \times k_l^3] / [4 \mu_l (T_{sat} - T_w) \times 1]$  so much  $W/m^2 \cdot ^\circ C$ .

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Hence, the recommended expression for vertical surfaces in laminar flow is:-

$$N_{Nu} = \frac{hL}{k_l} = 1.13 \left[ \frac{g \rho_l (\rho_l - \rho_v) h_{fg} L^3}{\mu_l (T_{sat} - T_w) k_l} \right]^{1/4}$$

where,  $\rho_l$  = density of liquid,  $\text{kg} / \text{m}^3$   
 $\rho_v$  = density of vapour,  $\text{kg} / \text{m}^3$ ;  $g = 9.8066 \text{ m} / \text{s}^2$ ;  $L$  = length of the surface in  $\text{m}$ ;  $\mu_l$  = viscosity of liquid in  $\text{Pa}\cdot\text{s}$ ;  $k_l$  = thermal conductivity of liquid in  $\text{W} / \text{m} \text{ } ^\circ\text{C}$ ;  $h_{fg}$  = latent heat of condensation in  $\text{J} / \text{kg} \text{ } ^\circ\text{C}$  at  $T_{sat}$ .  
All physical properties of the liquid except  $h_{fg}$  are evaluated at the film temperature  $T_f = (T_{sat} + T_w) / 2$

For laminar flow, experimental data show that this result is around 20% less than the experimental data for which a new proposal has come up from the researcher saying that Nusselt number can be related as this. Hence, the recommendation expression for the vertical surface in laminar flow is like this  $N_{Nu} = hL/k_l = 1.13 [g \rho_l \times (\rho_l - \rho_v) h_f g \times L^3] / [\mu_l (T_{sat} - T_w) \times k_l]^{1/4}$ .

So,  $h$  now is average, 'right',  $h$  is the average heat transfer coefficient not the local and that is related to Nusselt number as  $hL/k_l$  this is again not the in the wide number, it would have been the solid, but here it is the fluid  $k_l$  liquid. So,  $hL/k_l$  is the Nusselt number expressed like this, where obviously,  $\rho_l$  density of liquid in  $\text{kg}/\text{m}^3$   $\rho_v$  density of the vapor in  $\text{kg}/\text{m}^3$ ,  $g$  in  $9.8066 \text{ m}/\text{s}^2$ .

$L$  length of the surface in meter;  $\mu_l$  viscosity of liquid in  $\text{Pa}\cdot\text{s}$ ;  $k_l$  thermal conductivity of the liquid in  $\text{W}/\text{m}\cdot^\circ\text{C}$ ;  $h_f$   $g$  latent heat of condensation in  $\text{J}/\text{kg}\cdot^\circ\text{C}$  at  $T_{sat}$ . All physical properties of the liquid and  $h_f$  except  $h_f$   $g$  are evaluated at the film temperature  $T_f$  is  $(T_{sat} + T_w) / 2$  that is the average temperature of the saturated temperature of the vapor and the wall temperature, 'right'.

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Condensation on inclined surface:-

$$h_{av} = 0.943 \left[ \frac{g \rho_l (\rho_l - \rho_v) h_{fg} k_l^3 \sin \theta}{\mu_l (T_{sat} - T_w) L} \right]^{1/4} \quad W/m^2 \cdot ^\circ C$$

where,  $\theta$  = angle of the vertical surface with the horizontal.

Condensation on a horizontal tube:-

$$h_{av} = 0.725 \left[ \frac{g \rho_l (\rho_l - \rho_v) h_{fg} k_l^3 \sin \theta}{\mu_l (T_{sat} - T_w) D} \right]^{1/4} \quad W/m^2 \cdot ^\circ C$$

where, D is the outside diameter of the tube in m.

So, we can now say the condensation on inclined surface that can be written as  $h_{average}$  is  $0.943 \{ [g \rho_l (\rho_l - \rho_v) h_{fg} k_l^3] \sin \theta / [\mu_l (T_{sat} - T_w) L] \}^{1/4}$  depending on whether what is the inclination to the power 1/4 in  $W/m^2$ , where  $\theta$  is the angle of vertical surface with the horizontal condensation on a horizontal tube, and condensation of course, horizontal surface that is the angle.

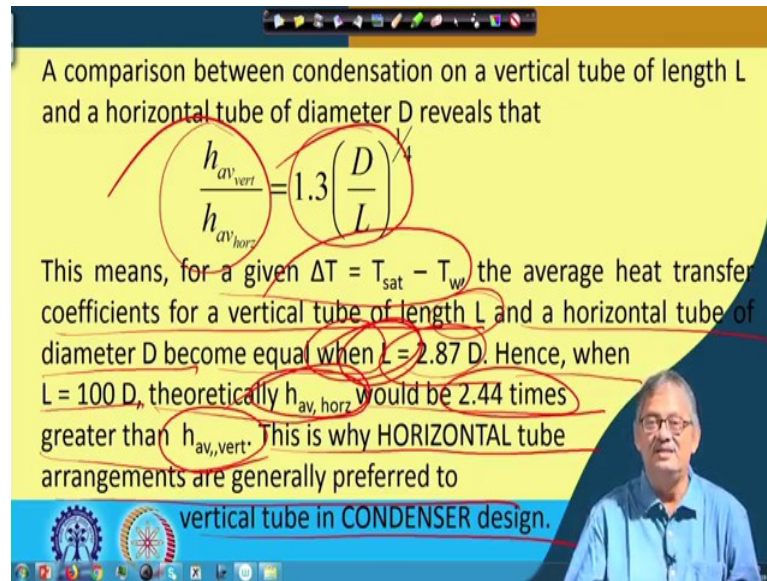
Now, condensation on the horizontal T will be  $h_{average}$  is  $0.725 \{ [g \rho_l (\rho_l - \rho_v) h_{fg} k_l^3] \sin \theta / [\mu_l (T_{sat} - T_w) D] \}^{1/4}$  so much  $W/m^2$ , where D is the outside diameter of the tube in meter.

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A comparison between condensation on a vertical tube of length  $L$  and a horizontal tube of diameter  $D$  reveals that

$$\frac{h_{av,vert}}{h_{av,horz}} = 1.3 \left( \frac{D}{L} \right)^{1/4}$$

This means, for a given  $\Delta T = T_{sat} - T_w$  the average heat transfer coefficients for a vertical tube of length  $L$  and a horizontal tube of diameter  $D$  become equal when  $L = 2.87 D$ . Hence, when  $L = 100 D$ , theoretically  $h_{av,horz}$  would be 2.44 times greater than  $h_{av,vert}$ . This is why HORIZONTAL tube arrangements are generally preferred to vertical tube in CONDENSER design.



And since our time is very low we can say that  $h_{av}$  vertical /  $h_{av}$  horizontal is  $1.3 (D/L)^{1/4}$ . So, this means, that  $\Delta T$  that is  $T_{saturation} - T_{wall}$  there is the average heat transfer coefficient for vertical tube of length  $L$  and horizontal tube of diameter  $D$  become equal only when  $L = 2.7$  time 2.8 times  $D$ .

So, when  $L = 100 D$  theoretically  $h_{av}$  horizontal would be 2.44 times greater than that of vertical edge. So, this is why horizontal tube arrangements are generally preferred over vertical in the condensers, 'right'.

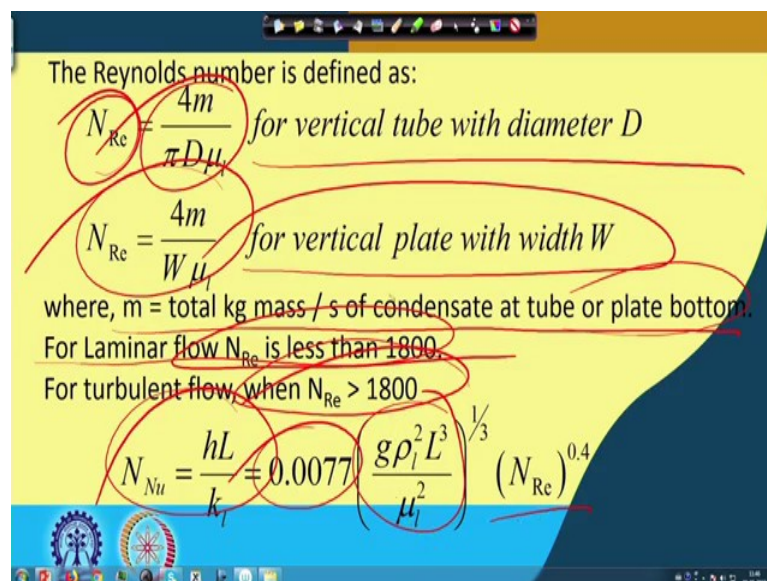
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The Reynolds number is defined as:

$$N_{Re} = \frac{4m}{\pi D \mu_l} \text{ for vertical tube with diameter } D$$

$$N_{Re} = \frac{4m}{W \mu_l} \text{ for vertical plate with width } W$$

where,  $m$  = total kg mass / s of condensate at tube or plate bottom.  
 For Laminar flow  $N_{Re}$  is less than 1800.  
 For turbulent flow, when  $N_{Re} > 1800$

$$N_{Nu} = \frac{hL}{k_f} = 0.0077 \left( \frac{g \rho_l^2 L^3}{\mu_l^2} \right)^{1/3} (N_{Re})^{0.4}$$


So, hopefully we have come to the end of the class. And we say some more here also a little just for few seconds that is  $N_{Re} = 4m / (\pi D \mu_i)$ , for vertical tube with diameter  $D$ .  $N_{Re}$  is  $4m / (W \mu_i)$  for vertical plate with width  $W$ . And  $m$  is the total kg mass / s of the condensate active or plate bottom. For laminar flow  $N_{Re} < 1800$ ; for turbulent flow  $N_{Re} > 1800$ . So, Nusselt number we can write  $0.0077 (g \rho_i^2 L^3 / \mu_i^2)^{1/3}$  and  $N_{Re}^{0.4}$ . So, with this we come to the end of the condensation.

And we thank you.